

**SECTION 9
SUPPLEMENTS**

OPTIONAL EQUIPMENT SUPPLEMENTS

Information contained in the following supplements applies only when the related equipment is installed.

CONTENTS

	Page
Police Version	9-3.1
Mariner/Mariner II	9-4.1
ADS-B Equipment	9-9.1
Optional Avionics	9-10.1

Supplements 1, 2, 5, 6, 7, and 8 are obsolete.

NON-U.S. SUPPLEMENTS

The following supplements contain additional information required by certain countries:

- Brazilian Supplement
- CIS Supplement
- IAC AR Supplement
- Ukrainian Supplement

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FAA APPROVED
R22 PILOT'S OPERATING HANDBOOK
SUPPLEMENT 3
TO
ROBINSON MODEL R22

This supplement must be attached to the FAA Approved Robinson R22 Pilot's Operating Handbook when the helicopter is configured with optional police version equipment.

The information contained herein supplements or supersedes the basic manual only in those areas listed in this Supplement. For limitations, procedures, and performance information not contained in this Supplement, consult the basic R22 Pilot's Operating Handbook.

FAA APPROVED *James M. Bugbee*
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FAA, Western Aircraft Certification Field Office
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DATE *March 27, 1984*

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9-3.3	04 Nov 1994
9-3.4	04 Nov 1994
9-3.5	04 Nov 1994
9-3.6	04 Nov 1994

REVISIONS
APPROVED BY *Donald Armstrong*
Manager, Flight Test Branch ANM-160L
Federal Aviation Administration, LAACO
Transport Airplane Directorate

DATE *Nov. 4, 1994*

SECTION 9
R22

POLICE VERSION

INTRODUCTION

This supplement contains the changes and additional data applicable when the R22 helicopter is equipped with Police Version options.

SECTION 2: Limitations

VFR operation at night is permitted with either landing lights or searchlight operable. Instrument and anti-collision lights must be installed and operable.

SECTION 3: Emergency Procedures

No change.

SECTION 4: Normal Procedures

Additions to preflight check.

Add to number 10, fuselage left side.

PA Speaker secure

Searchlight secure

SECTION 5: Performance

No change.

SECTION 6: Weight & Balance

No change.

SECTION 7: Systems Description

The following optional equipment may be installed:

Airframe

Cabin doors with bubble windows to enhance downward and rearward visibility replace the standard doors.

Electrical System

A 14-volt, 70-ampere alternator replaces the standard 60-ampere unit.

Searchlight Installation

The searchlight installation consists of two lamps suspended below the cabin chin on a post which extends through the left hand side of the chin and cabin floor. The searchlight is controlled by means of a grip and switch(es) mounted on the post allowing operation by the passenger or pilot. Power is supplied to the searchlight through its own 20 amp circuit breaker and power relay. The searchlight will operate only when the landing lights are off. A friction control is located on the post just above the floor.

Two searchlight installations are available: one manufactured by Collins Dynamics and the other by Peak Beam Systems. The Collins Dynamics unit uses incandescent quartz halogen lamps controlled by a single on-off toggle switch on the grip. The Peak Beam Systems unit uses xenon arc lamps controlled by four momentary toggle switches on the grip. These switches control on-off, strobe (rapidly flashing) mode, and beam spread from 1° spot to 40° flood.

CAUTION

When operating the searchlight with the left seat cyclic installed, caution should be exercised to prevent interference with the cyclic control.

SECTION 7: Systems Description (Cont'd)

PA System

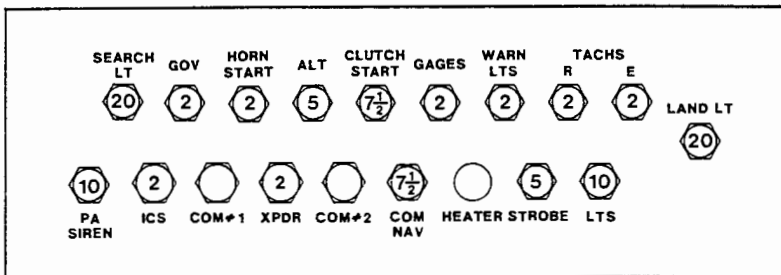
A 100-watt Whelen PA speaker and siren is located on the aft, left landing gear strut. The PA system control panel is located on the instrument panel. It allows the pilot or observer to select either PA, radio, yelp, or siren for broadcast through the PA speaker.

Remote Radio Transmit Switch

A hand-held remote radio transmit switch plugs into the receptacle on the front of the left seat. This switch allows the left seat occupant to transmit when the left cyclic is removed.

Circuit Breaker Panel

The circuit breaker panel is shown in Figure 1.



**FIGURE 1
CIRCUIT BREAKER PANEL**

SECTION 8: Handling and Maintenance

Searchlight Installation

To remove the searchlight from the helicopter, use the following procedure:

1. Loosen but do not remove two 1/4 in. diameter bolts in clamp on bottom end of post.
2. Remove clevis pin connecting lower end of tube through center of post to arms between lamps.
3. Disconnect wires from lamps at connectors on lamp housings.
4. Pull ring on spring loaded pin in post clamp and slide clamp with lamps off post.
5. Remove four screws holding post flange to cabin floor. Pull post up a few inches and disconnect wires at post connector. Before releasing, o-ring around the connector must be moved clear of retaining hooks. Pull post up and out of floor.
6. Cover connector under floor with tape and install sheet metal cover provided over hole in floor.

Installation is the reverse except:

1. Be sure to position o-ring on post wire connector under retaining hooks.
2. The letters "FWD" on post flange must face forward.
3. Be sure spring loaded pin in post clamp pops into place and torque 1/4 in. diameter bolts to 90 in.-lb (includes nut self-locking torque).

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FAA APPROVED
R22 PILOT'S OPERATING HANDBOOK

SUPPLEMENT 4
R22 MARINER/MARINER II

This supplement must be attached to the FAA Approved Robinson R22 Pilot's Operating Handbook when the helicopter is configured in the Mariner or Mariner II version.

The information contained herein supplements or supersedes the basic manual only in those areas listed in this Supplement. For limitations, procedures, and performance information not contained in this Supplement, consult the basic R22 Pilot's Operating Handbook.

FAA APPROVED Frank J. Hoerman
Supervisor, Flight Test Section
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Northwest Mountain Region

DATE Sept 7, 1985

LOG OF REVISIONS

Page No.	Date	Page No.	Date
9-4.1	13 Oct 00	9-4.8	13 Oct 00
9-4.2	13 Oct 00	9-4.9	13 Oct 00
9-4.3	13 Oct 00	9-4.10	13 Oct 00
9-4.4	13 Oct 00	9-4.11	13 Oct 00
9-4.5	13 Oct 00	9-4.12	13 Oct 00
9-4.6	13 Oct 00	9-4.13	13 Oct 00
9-4.7	13 Oct 00		

REVISIONS
APPROVED BY Donald C. ...
Manager, Flight Test Branch, ANM-160L
Federal Aviation Administration, LAACO
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DATE Oct-13, 2000

SECTION 1: GENERAL

INTRODUCTION

This supplement contains the changes and additional data applicable for R22 Mariner and Mariner II helicopters. R22 Mariner/Mariner II helicopters are equipped with either float or standard landing gear, additional corrosion protection, and a modified horizontal stabilizer angle of incidence. Except as noted in this supplement, information in the basic manual pertaining to the R22 Beta is applicable to the R22 Mariner and information pertaining to the Beta II is applicable to the Mariner II.

Float landing gear consists of inflated floats attached to the skid tubes, and skid extensions to support the aft portion of the float tubes. When float landing gear is installed, an additional stabilizer is installed at the bottom of the lower vertical stabilizer for improved controllability. Configuration changes between float and standard landing gear are permitted.

SECTION 2: LIMITATIONS

AIRSPPEED LIMITATIONS

NEVER EXCEED AIRSPEED (V_{ne}) WITH FLOATS

Power-On: 95 KIAS up to 3000 ft density altitude.

Power-Off: 80 KIAS up to 7500 ft density altitude.

Above these density altitudes, see placard on page 9-4.4.

CENTER OF GRAVITY (CG) LIMITS

See figure on page 9-4.3. Datum line is 100 inches forward of main rotor shaft centerline.

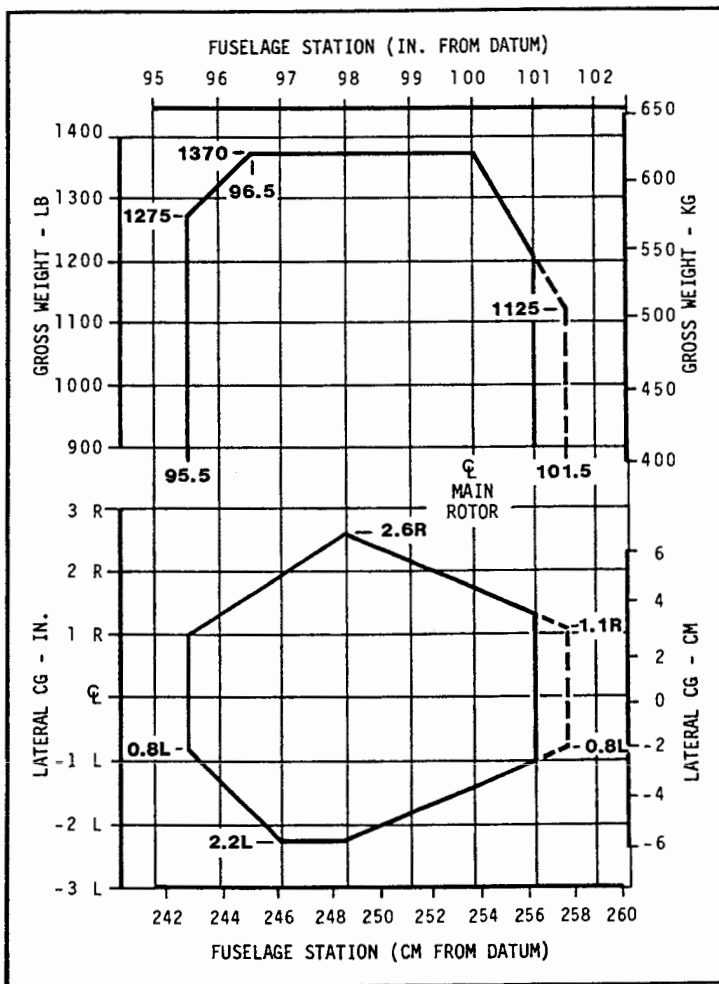
KINDS OF OPERATION LIMITATIONS

Only VFR day is approved when floats are installed.

SECTION 2: LIMITATIONS (cont'd)

WITH FLOATS _____

WITHOUT FLOATS - - - - -



R22 MARINER/MARINER II
 CENTER OF GRAVITY LIMITS

SECTION 2: LIMITATIONS (cont'd)

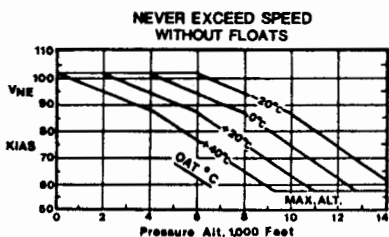
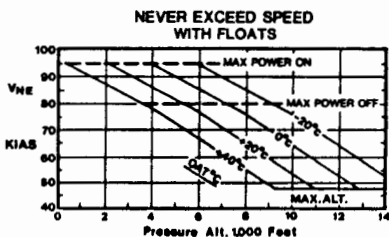
PLACARDS

In clear view of pilot:

THIS ROTORCRAFT IS APPROVED FOR
DAY AND NIGHT VFR OPERATIONS
WITHOUT FLOATS INSTALLED
OR
DAY VFR OPERATIONS ONLY
WITH FLOATS INSTALLED

In clear view of pilot in flight:

R22 Mariner



R22 Mariner II

**NEVER EXCEED SPEED - KIAS
WITH FLOATS**

PRESS	OAT - °C						
ALT-FT	-20	-10	0	10	20	30	40
SL							
2000	95				94	91	88
4000			94	90	86	83	79
6000		90	86	82	78	74	69
8000	86	82	77	72	67	62	57
10000	77	72	66	61	56	51	
12000	66	60	55				
14000	54						
NO FLIGHT							
FOR AUTOROTATION, USE 80 KIAS ABOVE HEAVY LINE							

**NEVER EXCEED SPEED - KIAS
WITHOUT FLOATS**

PRESS	OAT - °C						
ALT-FT	-20	-10	0	10	20	30	40
SL							
2000	102					99	96
4000				98	94	91	87
6000		98	94	90	87	82	77
8000	94	90	86	80	75	69	64
10000	86	80	74	68	62	57	
12000	74	67	61				
14000	61						
NO FLIGHT							

Located on floats:

SEE PILOT'S HANDBOOK FOR
PROPER INFLATION PRESSURE

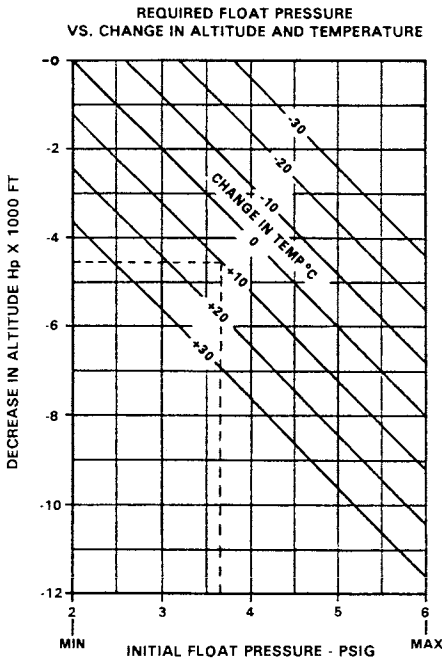
SECTION 2: LIMITATIONS (cont'd)

FLOAT PRESSURE LIMITS

Minimum Float Pressure: 2 psig (psi gage)

Maximum Float Pressure: 6 psig

The following graph shows minimum float pressure required for a decrease in altitude or temperature. An increase in either altitude or temperature can be ignored as pressure relief valves will limit pressure to 6 psig.



CAUTION

Failure to maintain required pressure can result in inadequate buoyancy or in-flight instability.

SAMPLE CALCULATION:

	<u>Pressure</u>	<u>Altitude</u>	<u>Temp</u>
Conditions at destination:	1000 ft	15°C	
Initial Conditions:	5500 ft	5°C	
Subtract to obtain expected change in altitude and temp:	-4500 ft	+10°C	

Using graph, locate -4500 ft line, read across to +10°C line, then down for minimum initial float pressure required, approximately 3.7 psig.

SECTION 3: EMERGENCY PROCEDURES

GENERAL

Without floats, emergency procedures in the basic manual apply. The following procedures are applicable when floats are installed.

POWER FAILURE ABOVE 500 FEET AGL

Autorotation to Land: Same as non-float R22.

Autorotation to Water: Steps 1-7 same as non-float R22.

8. At about 8 feet AGL, apply forward cyclic to level helicopter and raise collective just before touchdown to cushion landing. Touchdown in slight nose high attitude with nose straight ahead.
9. Maintain cyclic in touchdown position and hold collective up until forward motion has stopped. See caution below.

POWER FAILURE BETWEEN 8 AND 500 FEET AGL

Autorotation to Land: Same as non-float R22.

Autorotation to Water: Steps 1-4 same as non-float R22.

5. At about 8 feet AGL, apply forward cyclic to level helicopter and raise collective just before touchdown to cushion landing. Touchdown in slight nose high attitude with nose straight ahead.
6. Maintain cyclic in touchdown position and hold collective up until forward motion has stopped.

CAUTION

Lowering collective or applying forward cyclic while helicopter is moving forward on water can cause floats to submerge and helicopter to nose over.

DITCHING

Not applicable when floats are installed.

SECTION 4: NORMAL PROCEDURES

DAILY OR PREFLIGHT CHECKS

14. Inflatable Floats

- Float Pressure Check (See Section 2)
- Float Condition Check

CAUTION

Helicopters equipped with inflated floats have an adverse roll characteristic; when sideslipping nose left or right the helicopter will tend to roll in the opposite direction. To avoid adverse roll, keep helicopter trimmed with zero sideslip and exercise extreme caution when performing simulated power failures.

OPERATION ON WATER

Maximum recommended water taxi speed is 5 knots. Some application of collective is required.

CAUTION

If starting or stopping rotor on water, ensure area is clear as helicopter can rotate one complete turn while tail rotor RPM is low.

CAUTION

To avoid contacting rotor teeter stops, do not apply full cyclic control during water taxi.

NOTE

Safe operation on water has been demonstrated in waves up to 12 inches (trough to crest).

SECTION 4: NORMAL PROCEDURES (cont'd)

PRACTICE AUTOROTATION - WITH GROUND CONTACT

Same as non-float R22 except rear of floats may be scuffed on pavement during touchdown. Protective tape should be applied to bottom of floats and inspected for wear after touchdowns.

PRACTICE AUTOROTATION - WITH WATER CONTACT

Same as practice autorotation with ground contact in basic manual except touch down in slight nose high attitude with nose straight ahead. Maintain cyclic in touchdown position and collective up until forward motion has stopped.

CAUTION

Lowering collective or applying forward cyclic while helicopter is moving forward on water can cause floats to submerge and helicopter to nose over.

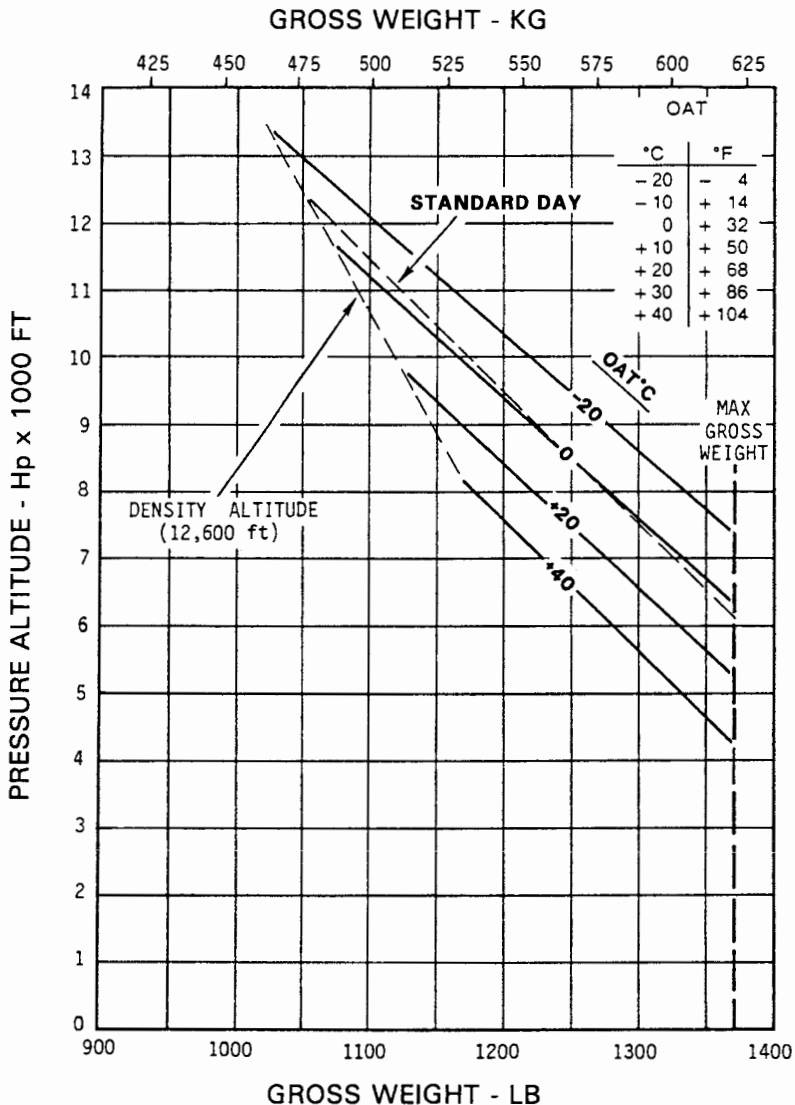
SECTION 5: PERFORMANCE

HOVER PERFORMANCE

In-ground-effect (IGE) and out-of-ground-effect (OGE) hover performance charts for the R22 Mariner are on pages 9-4.9 and 9-4.10.

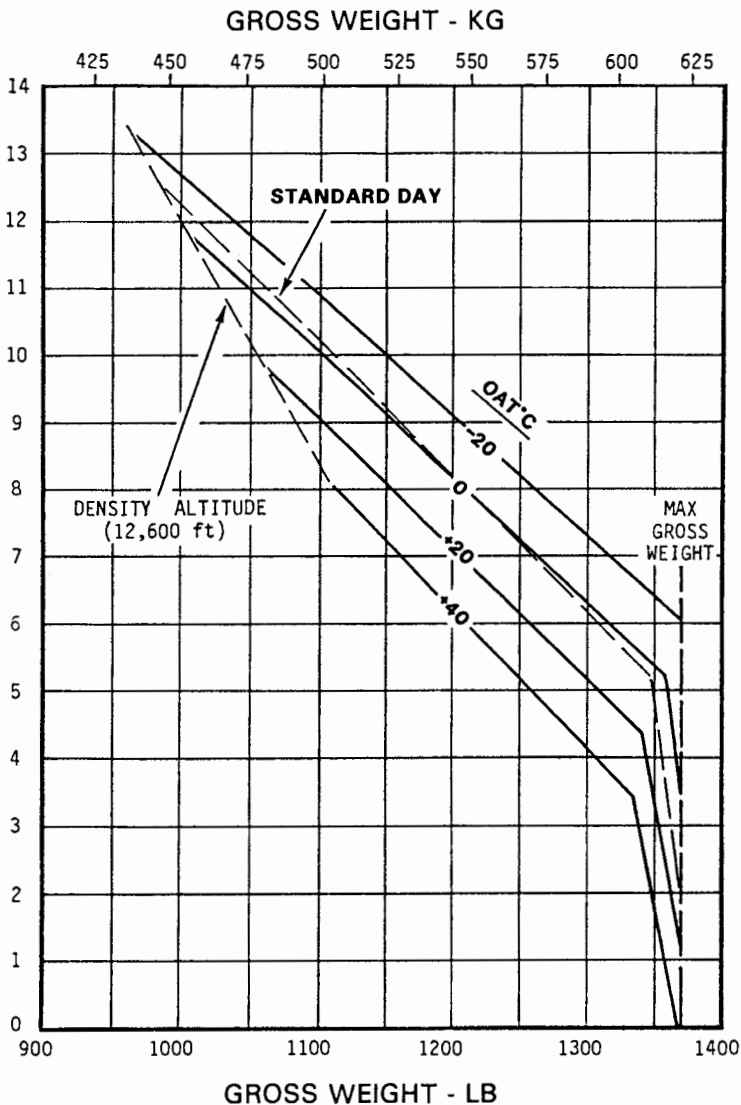
R22 Mariner II hover performance is equivalent to R22 Beta II hover performance.

**IN GROUND EFFECT AT 2 FOOT SKID CLEARANCE
FULL THROTTLE AND 104% RPM**



**R22 MARINER
O-320-B2C Engine
IGE HOVER CEILING VS. GROSS WEIGHT**

**OUT OF GROUND EFFECT
TAKEOFF POWER OR FULL THROTTLE AT 104% RPM**



**R22 MARINER
O-320-B2C Engine
5 Minute Takeoff Rating
OGE HOVER CEILING VS. GROSS WEIGHT**

SECTION 6: WEIGHT AND BALANCE

CAUTION

When changing from float-equipped to non-float configuration or vice versa, weight and balance for basic helicopter must be revised and autorotation RPM readjusted per R22 Maintenance Manual.

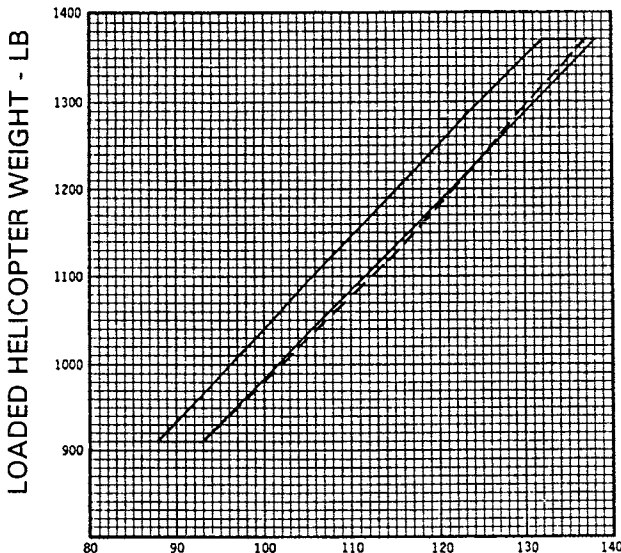
WEIGHT AND BALANCE RECORD

Basic empty weight and CG in float and non-float configurations is included in the Weight and Balance Data provided with the helicopter. Modifications are to be recorded on page 6-5 of the basic manual.

LOADING INSTRUCTIONS

Determine loaded helicopter weights and moments per page 6-7. These values must fall within the Weight and Moment envelope shown below.

WITHOUT FLOATS **WITH FLOATS**



LOADED HELICOPTER MOMENT - 1000 inch-lbs.

**R22 MARINER/MARINER II
Allowable CG Moment vs. Gross Weight Envelope**

SECTION 7: SYSTEMS DESCRIPTION

No change.

SECTION 8: HANDLING, SERVICING AND MAINTENANCE

GROUND HANDLING

When floats are installed, special ground handling wheels are required. Refer to the R22 Maintenance Manual for wheel installation and removal procedures.

FLOAT TUBES

To promote long float tube life:

1. Do not inflate floats to a higher pressure than required in the Limitations section.
2. Do not arbitrarily inflate floats to the relief valve pressure.
- 3 Reduce pressure in floats if solar heating is causing an excessive pressure buildup.
4. Do not allow floats to sit uninflated for long periods of time, always maintain some pressure to keep shape when not in use.

CAUTION

When inflating chambers individually (without the use of a manifold), increase pressure in each chamber in 0.5 psig increments.

SECTION 10: SAFETY TIPS

Flight characteristics and handling qualities of a helicopter with inflated floats are more critical than those for helicopters with conventional landing gear. For instance, helicopters with floats installed have an adverse roll characteristic, i.e. when sideslipping nose right or left, the helicopter will tend to roll in the opposite direction out of the turn. This could be extremely dangerous if a student or instructor failed to put in right pedal or put in the wrong pedal during a simulated power failure. Also, aerodynamic lift produced by the floats makes both RPM and longitudinal attitude control more difficult during autorotation entries. Helicopters with floats installed are also more gust sensitive and therefore more difficult to fly in turbulence.


For these reasons, it is strongly recommended that floats be removed and standard gear installed when the helicopter is being used for primary flight instruction. When floats are installed, pilots must keep the helicopter trimmed with zero sideslip and exercise extreme caution when performing simulated power failures.

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FAA APPROVED
R22 PILOT'S OPERATING HANDBOOK
ADS-B EQUIPMENT SUPPLEMENT

This supplement must be included in the FAA-approved Pilot's Operating Handbook when ADS-B equipment is installed.

The information contained herein supplements or supersedes the basic manual only in those areas listed in this supplement. For limitations, procedures, and performance information not contained in this supplement, consult the basic Pilot's Operating Handbook.

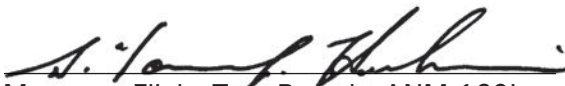
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Federal Aviation Administration, LAACO
Transport Airplane Directorate

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9-9.3*	26 Oct 16	9-9.6*	26 Oct 16

* Manufacturer's data, not FAA approved.

REVISIONS
APPROVED BY: 
Manager, Flight Test Branch, ANM-160L
Federal Aviation Administration, LAACO
Transport Airplane Directorate

DATE: October 26, 2016

SECTION 1: GENERAL

INTRODUCTION

This supplement contains the changes and additional data applicable when Automatic Dependent Surveillance-Broadcast (ADS-B) equipment is installed.

ADS-B is divided into two categories – ADS-B “Out” and ADS-B “In”.

ADS-B Out equipment transmits information to air traffic control to supplement radar/transponder information. The supplemental information allows optimization of flight plan routes and aircraft spacing.

ADS-B Out equipment may be required for operation in certain airspace. The R22 ADS-B Out installation has been shown to meet the requirements of 14 CFR § 91.227.

NOTE

The R22 ADS-B Out system operates on frequency 1090 MHz. This frequency is also accepted for ADS-B Out equipment in most countries outside the United States.

The ADS-B Out equipment consists of either a GPS receiver connected to the transponder or a transponder with built in GPS. The transponder has ADS-B broadcast capability and broadcasts GPS position as well as additional pre-programmed information such as aircraft identification and type to air traffic control.

ADS-B In equipment receives traffic information from other ADS-B equipped aircraft. ADS-B In equipment may also receive additional traffic information and weather information from ground stations. The additional traffic and weather information from ground stations is only available in the United States.

SECTION 1: GENERAL (cont'd)

INTRODUCTION (cont'd)

The ADS-B In equipment consists of a receiver built in to the transponder and a suitable display. Refer to receiver and display manufacturers documentation for operation of ADS-B In equipment.

The R22 may be equipped with only ADS-B Out or with both ADS-B Out and ADS-B In.

SECTION 2: LIMITATIONS

PLACARDS

On transponder when ADS-B Out equipment is installed:

ADS-B OUT INSTALLED

SECTION 3: EMERGENCY PROCEDURES

No change.

SECTION 4: NORMAL PROCEDURES

ADS-B SYSTEM OPERATION

ADS-B system operation is mostly automatic and requires little pilot action. The GPS (if separate from the transponder), transponder, and display (if installed) must all be powered and in normal operating modes for proper system function.

ADS-B OUT

The R22 ADS-B Out system is a single point of entry system. Mode 3/A codes, IDENT commands, and emergency codes are set on the transponder and are automatically incorporated in ADS-B Out broadcasts. The transponder should transition to ALT mode after takeoff for proper ADS-B Out broadcasts.

ADS-B Out broadcasts may be selected off by using menus associated with the transponder FUNC key.

NOTE

ADS-B Out may be required in certain airspace. Do not turn off ADS-B Out unless directed by air traffic control.

Malfunctions in the ADS-B Out system are annunciated by various messages on the transponder or display (refer to manufacturers' documentation).

SECTION 4: NORMAL PROCEDURES (cont'd)

ADS-B SYSTEM OPERATION (cont'd)

ADS-B IN

ADS-B In receiver is built in to the transponder and has no direct pilot interface.

ADS-B In data is sent from the receiver to a suitable display, often the primary GPS screen. The display may have dedicated traffic and weather views or may allow traffic and weather information to be overlaid on other data such as moving maps. Warnings such as traffic conflicts may also appear on the display. Refer to receiver and display manufacturers' documentation.

SECTION 5: PERFORMANCE

No change.

SECTION 6: WEIGHT AND BALANCE

No change.

SECTION 7: SYSTEM DESCRIPTION

ADS-B SYSTEM

The ADS-B Out system consists of either a GPS receiver connected to the transponder or a transponder with built-in GPS. The transponder broadcasts the aircraft's position, identification, and certain other parameters to air traffic control. ADS-B data is broadcast via the Extended Squitter (ES) feature of the transponder on a frequency of 1090 MHz. Note that change of aircraft registration may require update of preprogrammed parameters by qualified maintenance personnel.

Most of the data required for ADS-B broadcast such as aircraft type, ICAO address, and call sign are pre-programmed at installation. Flight-specific data such as Mode 3/A code and IDENT are entered using the transponder controls. The transponder uses these codes simultaneously for standard transponder as well as ADS-B broadcasts. There is no need to make a second code entry or to enter a code more than once. This is known as a "single point of entry" ADS-B system.

The ADS-B In system consists of a receiver built in to the transponder and a suitable display. The receiver receives both approved US ADS-B Frequencies (978 MHz and 1090 MHz).

SECTION 8: HANDLING, SERVICING AND MAINTENANCE

No change.

**FAA APPROVED
R22 PILOT'S OPERATING HANDBOOK
OPTIONAL AVIONICS SUPPLEMENT**

This supplement must be included in the FAA-approved Pilot's Operating Handbook when certain factory-supplied optional avionics are installed.

Information contained herein supplements or supersedes the basic manual only in those areas listed in this supplement. For limitations, procedures, and performance information not contained in this supplement, consult the basic Pilot's Operating Handbook.

APPROVED BY: Nevada Jo Ryan Digitally signed by Nevada Jo Ryan
Date: 2019.06.20 10:08:15 -07'00'
Manager, West Flight Test Section, AIR-716
Federal Aviation Administration
Los Angeles, CA

DATE: 20 JUN 2019

LOG OF PAGES

Page No.	Date	Page No.	Date
9-10.1	20 JUN 19	9-10.4*	20 JUN 19
9-10.2*	20 JUN 19	9-10.5*	20 JUN 19
9-10.3	20 JUN 19		

* Manufacturer's data, not FAA approved.

SECTION 1: GENERAL

INTRODUCTION

This supplement provides additional information for certain avionics options. A set of manufacturers' instructions for all installed avionics is provided with each new helicopter.

The following equipment is addressed in this supplement:

- Aspen Avionics EFD 1000H PFD

NOTE

For the Robinson Primary Flight Display (PFD) installation, the airspeed indicator, altimeter, compass, tachometer, and engine instruments are retained. Pilots should use the traditional instruments as primary unless fully familiar with the installed avionics.

SECTION 2: LIMITATIONS	No change.
SECTION 3: EMERGENCY PROCEDURES	No change.
SECTION 4: NORMAL PROCEDURES	No change.
SECTION 5: PERFORMANCE	No change.
SECTION 6: WEIGHT AND BALANCE	No change.
SECTION 7: SYSTEMS DESCRIPTION	See below.
SECTION 8: HANDLING AND MAINTENANCE	
No change.	

SECTION 7: SYSTEMS DESCRIPTION

ASPEN EFD 1000H PFD

The Aspen Electronic Flight Display (EFD) 1000H is a Primary Flight Display (PFD) optimized for helicopter use. It is available in a "Pilot" (basic) version or "Pro" (with more advanced navigation features) version.

A typical instrument panel is illustrated on the following page.

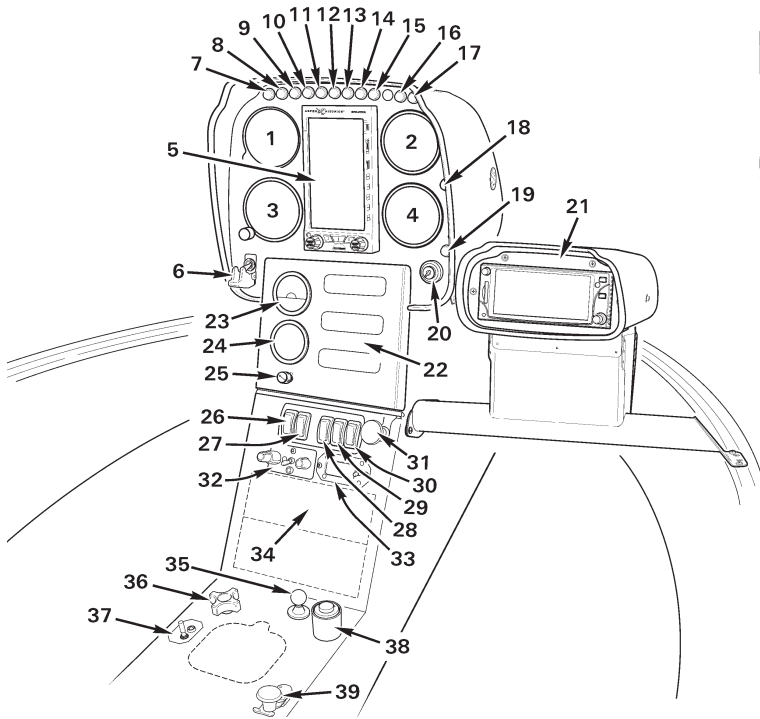
The manufacturer's document for the EFD 1000H is:

Title	Document No.
<i>Aspen Avionics Evolution Flight Display EFD 1000H PFD Pilot's Guide</i>	091-00012-001

NOTE

A Robinson part no. D327-4 light filter may be used to reduce reflections in the windshield at night. The light filter is installed by clipping it to the front of the display. Filter use is at pilot discretion.

SECTION 7: SYSTEMS DESCRIPTION (cont'd)



- | | |
|-----------------------------|----------------------------------|
| 1. VERTICAL SPEED INDICATOR | 21. PILOT'S SIDE CONSOLE (OPT'L) |
| 2. ENGINE AND ROTOR TACHS | 22. ENGINE INSTRUMENTS |
| 3. ALTIMETER | 23. CARBURETOR AIR TEMP |
| 4. MANIFOLD PRESSURE GAGE | 24. CLOCK |
| 5. PRIMARY FLIGHT DISPLAY | 25. PANEL LIGHTS DIMMER |
| 6. CLUTCH ACTUATOR SWITCH | 26. NAVIGATION LIGHTS SWITCH |
| 7. CLUTCH ACTUATOR LIGHT | 27. ANTI-COLLISION LIGHT SWITCH |
| 8. M.R. GEARBOX TEMP LIGHT | 28. AVIONICS MASTER SWITCH |
| 9. M.R. GEARBOX CHIP LIGHT | 29. ALTERNATOR SWITCH |
| 10. CARBON MONOXIDE LIGHT | 30. BATTERY SWITCH |
| 11. STARTER-ON LIGHT | 31. CABIN AIR |
| 12. T.R. GEARBOX CHIP LIGHT | 32. INTERCOM |
| 13. LOW FUEL LIGHT | 33. OUTSIDE AIR TEMP/VOLTMETER |
| 14. LOW RPM LIGHT | 34. AVIONICS STACK |
| 15. ALT LOW VOLTAGE LIGHT | 35. CYCLIC RIGHT TRIM |
| 16. OIL PRESSURE LIGHT | 36. CYCLIC FRICTION |
| 17. GOVERNOR-OFF LIGHT | 37. ELT SWITCH (OPT'L) |
| 18. FULL THROTTLE LIGHT | 38. MIXTURE CONTROL |
| 19. ROTOR BRAKE LIGHT | 39. CARBURETOR HEAT |
| 20. IGNITION SWITCH | |

**OPTIONAL INSTRUMENT PANEL WITH
ASPEN EFD 1000H PFD**

(Exact panel configuration may vary with optional equipment and date of helicopter manufacture.)

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