

**SECTION 4**

**NORMAL PROCEDURES**

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**SECTION 4**

**NORMAL PROCEDURES**

**RECOMMENDED AIRSPEEDS**

Takeoff and Climb	60 KIAS
Maximum Rate of Climb ( $V_Y$ )	55 KIAS
Maximum Range	100 KIAS*
Maximum Cruise (Do not exceed except in smooth air, and then only with caution)	110 KIAS*
Significant Turbulence	60 to 70 KIAS
Landing Approach	60 KIAS
Autorotation	60 to 70 KIAS*

\* Certain conditions may require lower airspeed.  
See  $V_{ne}$  placard in Section 2.

**DAILY OR PREFLIGHT CHECKS**

Remove all covers and tiedowns. Remove even small accumulations of frost, ice, or snow, especially from rotor blades. Check maintenance records to verify aircraft is airworthy. An 8-foot step ladder is recommended for preflight inspection of the main rotor; however, main rotor hub may be reached by first opening right rear seat and stepping on seat support and then stepping on deck below the aux fuel tank.

Check general condition of aircraft and verify no visible damage, fluid leakage, or abnormal wear. Verify no fretting at seams where parts are joined together. Fretting of aluminum parts produces a fine black powder while fretting of steel parts produces a reddish-brown or black residue. Verify Telatemp's show no temperature increase that cannot be attributed to a change in operating conditions (mechanics draw a reference line to the right of the highest temperature square which has darkened in operation). Verify torque stripes on critical fasteners are not broken or missing.

**DAILY OR PREFLIGHT CHECKS (cont'd)**

1. Upper Forward Cowl Doors - Right Side

- Battery switch . . . . . ON
- Oil pressure, alt, aux fuel pump lights . . . . . ON
- Warning light test switches . . . . . Push to test
- Fuel quantity . . . . . Check gages
- Battery switch . . . . . OFF
- Aux fuel tank quantity . . . . . Check
- Fuel filler cap . . . . . Tight
- Aux fuel tank . . . . . No leaks
- Fuel lines . . . . . No leaks
- Fuel tank sump, gascolator drains . . . . . Sample
- Gearbox oil . . . . . Full, no leaks
- Hydraulic system . . . . . Fluid full, no leaks
- Rotor brake . . . . . Actuation normal
- Flex coupling . . . . . No cracks, nuts tight
- Yoke flanges . . . . . No cracks
- Gearbox, hydraulic pump Telatemp . . . . . Normal
- Control rod ends . . . . . Free without looseness
- Steel tube frame . . . . . No cracks
- All fasteners . . . . . Tight
- Tail rotor control . . . . . No interference

2. Main Rotor

**CAUTION**

Do not pull down on blades to teeter rotor. To lower a blade, push up on opposite blade.

Blades . . . . . Clean and no damage/cracks

**CAUTION**

Verify erosion on lower surface of blades has not exposed skin-to-spar bond line. Reference Rotor Systems description in Section 7.

**DAILY OR PREFLIGHT CHECKS (cont'd)**

- 2. Main Rotor (cont'd)
  - Pitch change boots . . . . . No leaks
  - Main hinge bolts . . . . . Cotter pins installed
  - All rod ends . . . . . Free without looseness
  - Pitch link jam nuts . . . . . Tight
  - Pitch link safety wire . . . . . Secure
  - All fasteners . . . . . Tight
  - Swashplate scissors . . . . . No excessive looseness
  - Upper forward cowl doors . . . . . Latched
  
- 3. Lower Cowl Door - Right Side
  - Air box and duct . . . . . Secure
  - Engine sheet metal . . . . . No cracks
  - Fuel lines . . . . . No leaks
  - Oil lines . . . . . No leaks or chafing
  - Exhaust system . . . . . No cracks
  - Throttle linkage . . . . . Operable
  - Cowl door . . . . . Latched
  
- 4. Aft Cowl Door - Right Side
  - Oil cooler door . . . . . Check
  - V-belt condition . . . . . Check
  - V-belt slack . . . . . Check
  - Sprag clutch . . . . . No leaks
  - Upper bearing . . . . . No leaks
  - Telatemp - upper bearing . . . . . Normal
  - Sheave condition . . . . . Check
  - Flex coupling . . . . . No cracks, nuts tight
  - Yoke flanges . . . . . No cracks
  - Steel tube frame . . . . . No cracks
  - Tail rotor control . . . . . No interference
  - Tailcone attachment bolts . . . . . Check
  - Cowl door . . . . . Latched
  
- 5. Engine Rear
  - Cooling fan nut . . . . . Pin in line with marks
  - Cooling fan . . . . . No cracks
  - Fan scroll . . . . . No cracks
  - Tailpipe hanger . . . . . No cracks

**DAILY OR PREFLIGHT CHECKS (cont'd)**

- 6. Empennage
  - Tail surfaces . . . . . No cracks
  - Fasteners . . . . . Tight
  - Position light . . . . . Check
  - Tail rotor guard . . . . . No cracks
  
- 7. Tail Rotor
  - Gearbox Telatemp . . . . . Normal
  - Gearbox . . . . . Oil visible, no leaks
  - Blades . . . . . Clean and no damage/cracks
  - Pitch links . . . . . No looseness
  - Teeter bearings . . . . . Check condition
  - Teeter bearing bolt . . . . . Does not rotate
  - Control bellcrank . . . . . Free without looseness
  
- 8. Tailcone
  - Rivets . . . . . Tight
  - Skins . . . . . No cracks or dents
  - Strobe light condition . . . . . Check
  - Antenna . . . . . Check
  
- 9. Cowl Door - Left Side
  - Engine oil . . . . . 7-9 qts
  - Oil filter . . . . . Secure, no leaks
  - Battery and relay (if located here) . . . . . Secure
  - Steel tube frame . . . . . No cracks
  - Engine sheet metal . . . . . No cracks
  - Exhaust system . . . . . No cracks
  - Cowl door . . . . . Latched
  
- 10. Fuel tank (Main)
  - Quantity . . . . . Check
  - Filler cap . . . . . Tight
  - Leakage . . . . . None
  - Sump drain (non-bladder tank) . . . . . Sample

**DAILY OR PREFLIGHT CHECKS (cont'd)**

11. Fuselage Left Side

- Baggage compartments . . . . . Check
- Removable controls . . . . . Secure if installed
- Collective control . . . . . Clear
- Seat belts . . . . . Check condition and fastened
- Doors . . . . . Unlocked and latched
- Door hinge safety pins . . . . . Installed
- Landing gear . . . . . Check
- Ground handling wheel . . . . . Removed
- Position light . . . . . Check
- Static port . . . . . Clear

12. Nose Section

- Pitot tube . . . . . Clear
- Windshield condition and cleanliness . . . . . Check
- Fresh air vent . . . . . Clear
- Landing lights . . . . . Check

13. Fuselage Right Side

- Baggage compartments . . . . . Check
- Seat belts . . . . . Check condition and fastened
- Aft door . . . . . Unlocked and latched
- Door hinge safety pins . . . . . Installed
- Landing gear . . . . . Check
- Ground handling wheel . . . . . Removed
- Position light . . . . . Check
- Static port . . . . . Clear

14. Cabin Interior

- Loose articles . . . . . Removed or stowed
- Instruments, switches, and controls. . . . . Check condition
- Clock . . . . . Functioning
- Adjustable pedals . . . . . Pins secure

**CAUTION**

Remove left seat controls if person in that seat is not a rated helicopter pilot.

**DAILY OR PREFLIGHT CHECKS (cont'd)**

***CAUTION***

Fill baggage compartments under unoccupied seats to capacity before using baggage compartments under occupied seats. Avoid placing objects in compartment which could injure occupant if seat collapses during a hard landing.

***CAUTION***

Ensure all doors are unlocked before flight to allow rescue or exit in an emergency.

***CAUTION***

Shorter pilots may require cushion to obtain full travel of all controls. Verify aft cyclic travel is not restricted.

**BEFORE STARTING ENGINE**

- Seat belts . . . . . Fastened
- Fuel shut-off valve . . . . . ON
- Cyclic/collective friction . . . . . OFF
- Cyclic, collective, pedals . . . . . Full travel free
- Throttle . . . . . Full travel free
- Collective . . . . . Full down, friction ON
- Cyclic . . . . . Neutral, friction ON
- Pedals . . . . . Neutral
- Rotor brake . . . . . Disengaged
- Circuit breakers . . . . . In
- Landing lights . . . . . OFF
- Avionics switch (if installed) . . . . . OFF
- Clutch . . . . . Disengaged
- Altimeter . . . . . Set
- HYD and governor switches . . . . . ON



**ENGINE STARTING TIPS**

During prime, aux fuel pump warning light may remain illuminated momentarily. Continue prime 3 to 5 seconds after light extinguishes. If engine does not fire after 5 to 7 seconds of cranking, repeat priming sequence and reattempt start. If engine fails to start after three attempts, allow starter to cool ten minutes before next attempt.

If engine fires momentarily but dies before or while moving mixture to rich, pull mixture off, engage starter, and push mixture slowly rich while cranking.

**STARTING ENGINE AND RUN-UP**

Throttle . . . . . Closed  
Battery, strobe switches . . . . . ON  
Area . . . . . Clear  
Mixture . . . . . Rich  
Ignition switch . . . . . Prime, then Both  
Mixture . . . . . Pull OFF  
Starter . . . . . Engage until engine fires  
Mixture . . . . . Move full rich  
Mixture guard . . . . . Installed  
Starter-On light . . . . . Out  
Set engine RPM . . . . . 50 to 60%  
Clutch switch . . . . . Engaged  
Blades turning . . . . . Less than 5 seconds  
Alternator switch . . . . . ON  
Oil pressure within 30 seconds . . . . . 25 psi minimum  
Avionics, headsets . . . . . ON  
Wait for clutch light out . . . . . Circuit breakers in  
Warm-up RPM . . . . . 60 to 70%  
Engine gages . . . . . Green  
Mag drop at 75% RPM . . . . . 7% max in 2 seconds  
Sprag clutch check . . . . . Needles split  
Doors . . . . . Closed and latched  
Limit MAP chart . . . . . Check  
Cyclic/collective friction . . . . . OFF  
Hydraulic system . . . . . Check  
Governor On, increase throttle . . . . . RPM 101-102%  
Warning lights . . . . . Out  
Lift collective slightly, reduce RPM . . . . . Horn/Light at 97%

**STARTING ENGINE AND RUN-UP (cont'd)**

***CAUTION***

On slippery surfaces, be prepared to counter nose-right rotation with left pedal as governor increases RPM.

***NOTE***

For hydraulic system check, use small cyclic inputs. With hydraulics OFF, there should be approximately one half inch of freeplay before encountering control stiffness and feedback. With hydraulics ON, controls should be free with no feedback or uncommanded motion.

***NOTE***

Idle mixture and speed may require adjustment as conditions vary from sea level standard. Refer to R44 Maintenance Manual for idle adjustment procedure.

***NOTE***

Before takeoff, pilot should uncover one ear and listen for any unusual noise which may indicate impending failure of a bearing or other component.

**TAKEOFF PROCEDURE**

1. Verify doors latched, governor and hydraulics ON, and RPM stabilized at 101 to 102%.
2. Clear area. Slowly raise collective until aircraft is light on skids. Reposition cyclic as required for equilibrium, then gently lift aircraft into hover.
3. Check gages in green, lower nose, and accelerate to climb speed following profile shown by height-velocity diagram in Section 5. Avoid exceeding two inches MAP above IGE hover power to prevent excessive nose-down attitude. If RPM drops below 101%, lower collective.

## **CRUISE**

1. Verify RPM in green arc.
2. Set manifold pressure as desired with collective. Observe MAP and airspeed limits. Maximum recommended cruise speed is 110 KIAS.
3. Verify gages in green, warning lights out.

### ***CAUTION***

Do not exceed 110 KIAS except in smooth air, and then only with caution. In turbulence, use lower airspeed. If turbulence is significant or becomes uncomfortable for the pilot, use 60 to 70 KIAS.

### ***CAUTION***

In-flight leaning with engine mixture control is not allowed. Mixture must be full rich during flight.

### ***NOTE***

When loaded near aft CG limit, slight yaw oscillation during cruise can be stopped by applying a small amount of left pedal.

## **DOORS-OFF OPERATION**

Maximum airspeed with any door(s) off is 100 KIAS. Warn passengers to secure loose objects and to keep head and arms inside cabin to avoid high velocity airstream.

### ***CAUTION***

Ensure all seat belts are buckled during door-off flight. Rear seat bottoms may lift if not restrained and items in baggage compartments could be blown out.

### ***CAUTION***

Flight with left door(s) removed is not recommended. Loose objects exiting left doors may damage tail rotor.

**PRACTICE AUTOROTATION - POWER RECOVERY**

1. Lower collective to down stop and adjust throttle as required for small tachometer needle separation.

***CAUTION***

To avoid inadvertent engine stoppage, do not roll throttle to full idle. Roll throttle off smoothly only enough for a small visible needle split.

***NOTE***

Governor is inactive below 80% engine RPM regardless of governor switch position.

***NOTE***

When entering autorotation from above 6000 feet, reduce throttle slightly before lowering collective to prevent engine overspeed.

2. Adjust collective to keep rotor RPM in green arc and adjust throttle for small needle separation.
3. Keep airspeed 60 to 70 KIAS.
4. At about 40 feet AGL, begin cyclic flare to reduce rate of descent and forward speed.
5. At about 8 feet AGL, apply forward cyclic to level aircraft and raise collective to control descent. Add throttle if required to keep RPM in green arc.

***CAUTION***

Simulated engine failures require prompt lowering of collective to avoid dangerously low rotor RPM. Catastrophic rotor stall could occur if the rotor RPM ever drops below 80% plus 1% per 1000 feet of altitude.

**PRACTICE AUTOROTATION - WITH GROUND CONTACT**

If practice autorotations with ground contact are required for demonstration purposes, perform in same manner as power recovery autorotations except:

Prior to cyclic flare, roll throttle off into overtravel spring and hold against hard stop until autorotation is complete. (This prevents throttle correlator from adding power when collective is raised.)

Always contact ground with skids level and nose straight ahead.

***NOTE***

Have landing gear skid shoes inspected frequently when practicing autorotations with ground contact. Rapid wear of skid shoes may occur.

**HYDRAULICS-OFF TRAINING**

Hydraulic system failure may be simulated using the cyclic-mounted hydraulic switch.

***CAUTION***

With hydraulics switched OFF, controlling helicopter in a hover may be difficult due to control system feedback forces.

***CAUTION***

Before switching hydraulics from OFF to ON, relax force on cyclic and collective to avoid overcontrolling.

**DESCENT, APPROACH, AND LANDING**

1. Reduce power with collective as desired. Observe airspeed limits. Maximum recommended airspeed is 110 KIAS except in smooth air.

***CAUTION***

Do not initiate a descent with forward cyclic. This can produce a low-G condition. Always initiate a descent by lowering collective.

2. Make final approach into wind at lowest practical rate of descent with initial airspeed of 60 knots.
3. Reduce airspeed and altitude smoothly to hover. (Be sure rate of descent is less than 300 FPM before airspeed is reduced below 30 KIAS.)
4. From hover, lower collective gradually until ground contact.
5. After initial ground contact, lower collective to full down position.

***CAUTION***

When landing on a slope, return cyclic control to neutral before reducing rotor RPM.

***CAUTION***

Never leave helicopter flight controls unattended while engine is running.

***CAUTION***

Hold throttle closed if passenger is entering or exiting with engine running and left seat collective installed.

**SHUTDOWN PROCEDURE**

Collective down, RPM 60-70% . . . . . Friction ON  
Cyclic and pedals neutral . . . . . Friction ON  
CHT drop . . . . . Throttle closed  
Clutch switch . . . . . Disengage  
Wait 30 seconds . . . . . Mixture OFF  
Wait 30 seconds . . . . . Apply rotor brake  
Clutch light . . . . . Extinguishes  
Avionics, alt, battery, and ignition switches . . . . . OFF

***CAUTION***

Do not slow rotor by raising collective during shutdown. Blades may flap and strike tailcone.

***NOTE***

During idle and after engine shutdown, pilot should uncover one ear and listen for unusual noise which may indicate impending failure of a bearing or other component.

***NOTE***

HYD switch should be left ON for startup and shutdown to reduce possibility of unintentional hydraulics-off liftoff. Switch OFF only for pre-takeoff controls check or hydraulics-off training.

***NOTE***

Rotor brake should be left engaged after shutdown to disable starter buttons and reduce possibility of unintentional starter engagement.

**NOISE ABATEMENT**

To improve the quality of our environment and to dissuade overly restrictive ordinances against helicopters, it is imperative that every pilot minimize noise irritation to the public. Following are several techniques which should be employed when possible.

1. Avoid flying over outdoor assemblies of people. When this cannot be avoided, fly as high as practicable, preferably over 2000 feet AGL.
2. Avoid blade slap. Blade slap generally occurs at airspeeds below 100 KIAS. It can usually be avoided by maintaining 100 KIAS until rate of descent is over 1000 FPM, then using a fairly steep approach until airspeed is below 65 KIAS. With the right door vent open, the pilot can easily determine those flight conditions which produce blade slap and develop piloting techniques to eliminate or reduce it.
3. When departing from or approaching a landing site, avoid prolonged flight over noise sensitive areas. Always fly above 500 feet AGL and preferably above 1000 feet AGL.
4. Repetitive noise is far more irritating than a single occurrence. If you must fly over the same area more than once, vary your flight path to not overfly the same buildings each time.
5. When overflying populated areas, look ahead and select the least noise sensitive route.

**NOTE**

Above procedures do not apply where they would conflict with Air Traffic Control clearances or instructions or when, in the pilot's judgment, they would result in an unsafe flight path.