

SECTION 4

NORMAL PROCEDURES

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SECTION 4

NORMAL PROCEDURES

RECOMMENDED AIRSPEEDS

Takeoff and Climb	60 KIAS
Maximum Rate of Climb (V_Y)	55 KIAS
Maximum Range	100 KIAS*
Maximum Cruise (Do not exceed except in smooth air, and then only with caution)	110 KIAS*
Significant Turbulence	60 to 70 KIAS
Landing Approach	60 KIAS
Autorotation	60 to 70 KIAS*

* Certain conditions may require lower airspeed.
See V_{ne} placard in Section 2.

DAILY OR PREFLIGHT CHECKS

Remove ground handling wheels and all covers and tiedowns. Remove even small accumulations of frost, ice, or snow, especially from rotor blades. Check maintenance records to verify aircraft is airworthy. An 8-foot step ladder is recommended for preflight inspection of the main rotor; however, main rotor hub may be reached by first opening right rear seat and stepping on seat support and then stepping on deck below the aux fuel tank.

Check general condition of aircraft and verify no visible damage, fluid leakage, or abnormal wear. Verify no fretting at rivets and seams where parts are joined together. Fretting of aluminum parts produces a fine black powder while fretting of steel parts produces a reddish-brown or black residue. Verify Telatemp's show no temperature increase that cannot be attributed to a change in operating conditions (mechanics draw a reference line to the right of the highest temperature square which has darkened in operation). Verify torque stripes on critical fasteners are not broken or missing.

DAILY OR PREFLIGHT CHECKS (cont'd)

1. Upper Forward Cowl Doors – Right Side

- Battery switch ON
- Oil pressure, alt, aux fuel pump lights ON
- Warning light test switches Push to test
- Fuel quantity Check gages
- Battery switch OFF
- Aux fuel tank quantity Check
- Fuel filler cap Tight
- Aux fuel tank No leaks
- Fuel lines No leaks
- Fuel tank sump, gascolator drains Sample
- Gearbox oil Full, no leaks
- Hydraulic system Fluid full, no leaks
- Rotor brake Actuation normal
- Flex coupling No cracks, nuts secure
- Yoke flanges No cracks
- Gearbox, hydraulic pump Telatemp Normal
- Control rod ends Free without looseness
- Steel tube frame No cracks
- All fasteners Secure
- Tail rotor control No interference

2. Main Rotor

CAUTION

Do not pull down on blades to teeter rotor. To lower a blade, push up on opposite blade.

Blades Clean and no damage/cracks

CAUTION

Verify erosion on lower surface of blades has not exposed skin-to-spar bond line. Reference Rotor Systems description in Section 7.

DAILY OR PREFLIGHT CHECKS (cont'd)

2. Main Rotor (cont'd)

- Pitch change boots No leaks
- Main hinge bolts Cotter pins installed
- All rod ends Free without looseness
- All fasteners Secure
- Swashplate scissors No excessive looseness
- Upper forward cowl doors Latched

3. Lower Cowl Door – Right Side

- Air box and duct Secure
- Engine sheet metal No cracks
- Fuel lines No leaks
- Oil lines No leaks or chafing
- Exhaust system No cracks
- Throttle linkage Operable
- Cowl door Latched

4. Aft Cowl Door – Right Side

- Oil cooler door Check
- V-belt condition Check
- V-belt slack Check
- Sprag clutch No leaks
- Upper bearing No leaks
- Telatemp – upper bearing Normal
- Sheave condition Check
- Flex coupling No cracks, nuts secure
- Yoke flanges No cracks
- Steel tube frame No cracks
- Tail rotor control No interference
- Tailcone attachment bolts Check
- Cowl door Latched

5. Engine Rear

- Cooling fan nut Pin in line with marks
- Cooling fan No cracks
- Fan scroll No cracks
- Tailpipe hanger No cracks

DAILY OR PREFLIGHT CHECKS (cont'd)

6. Empennage

- Tail surfaces No cracks
- Fasteners Secure
- Position light Check
- Tail rotor guard No cracks

7. Tail Rotor

- Gearbox Telatemp Normal
- Gearbox Oil visible, no leaks
- Blades Clean and no damage/cracks
- Pitch links No looseness
- Teeter bearings Check condition
- Teeter bearing bolt Does not rotate
- Control bellcrank Free without looseness

8. Tailcone

- Skins No cracks or dents
- Strobe light condition Check
- Antenna Check

9. Cowl Door – Left Side

- Engine oil 7-9 qts
- Oil filter Secure, no leaks
- Battery and relay (if located here) Secure
- Steel tube frame No cracks
- Engine sheet metal No cracks
- Exhaust system No cracks
- Cowl door Latched

10. Main Fuel Tank

- Quantity Check
- Filler cap Tight
- Leakage None

DAILY OR PREFLIGHT CHECKS (cont'd)

11. Fuselage Left Side

- Baggage compartments Check
- Removable controls Secure if installed
- Collective controls Clear
- Seat belts Check condition and fastened
- Doors Unlocked and latched
- Door hinge safety pins Installed
- Landing gear Check
- Position light Check
- Static port Clear

12. Nose Section

- Pitot tube Clear
- Windshield condition and cleanliness Check
- Landing lights Check
- Yaw string Check

13. Fuselage Right Side

- Baggage compartments Check
- Seat belts Check condition and fastened
- Aft door Unlocked and latched
- Door hinge safety pins Installed
- Landing gear Check
- Position light Check
- Static port Clear

14. Cabin Interior

- Loose articles Removed or stowed
- Instruments, switches, and controls Check condition
- Clock Functioning
- Adjustable pedals Pins secure

CAUTION

Remove left seat controls if person in that seat is not a rated helicopter pilot.

DAILY OR PREFLIGHT CHECKS (cont'd)

CAUTION

Fill baggage compartments under unoccupied seats to capacity before using baggage compartments under occupied seats. Avoid placing objects in compartment which could injure occupant if seat collapses during a hard landing.

CAUTION

Ensure all doors are unlocked before flight to allow rescue or exit in an emergency.

CAUTION

Shorter pilots may require cushion to obtain full travel of all controls. Verify aft cyclic travel is not restricted.

BEFORE STARTING ENGINE

- Seat belts Fastened
- Fuel shut-off valve ON
- Cyclic/collective friction OFF
- Cyclic, collective, pedals Full travel free
- Throttle Full travel free
- Collective Full down, friction ON
- Cyclic Neutral, friction ON
- Pedals Neutral
- Rotor brake Disengaged
- Circuit breakers In
- Landing lights OFF
- Avionics switch (if installed) OFF
- Clutch Disengaged
- Altimeter Set
- HYD and governor switches ON

ENGINE STARTING TIPS

During prime, aux fuel pump warning light may remain illuminated momentarily. Continue prime 3 to 5 seconds after light extinguishes. If engine does not fire after 5 to 7 seconds of cranking, repeat priming sequence and reattempt start. If engine fails to start after three attempts, allow starter to cool ten minutes before next attempt.

If engine fires momentarily but dies before or while moving mixture to rich, pull mixture off, engage starter, and push mixture slowly rich while cranking.

STARTING ENGINE AND RUN-UP

Throttle Closed
Battery, strobe switches ON
Area Clear
Mixture Rich
Ignition switch Prime, then Both
Mixture Pull OFF
Starter Engage until engine fires
Mixture Move full rich
Mixture guard Installed
Starter-On light Out
Set engine RPM 50 to 60%
Clutch switch Engaged
Blades turning Less than 5 seconds
Alternator switch ON
Oil pressure within 30 seconds 25 psi minimum
Avionics, headsets ON
Wait for clutch light out Circuit breakers in
Warm-up RPM 60 to 70%
Engine gages Green
Mag drop at 75% RPM 7% max in 2 seconds
Sprag clutch check Needles split
Doors Closed and latched
Limit MAP chart Check
Cyclic/collective friction OFF
Hydraulic system Check
Governor On, increase throttle RPM 101-102%
Warning lights Out
Lift collective slightly, reduce RPM Horn/Light at 97%

STARTING ENGINE AND RUN-UP (cont'd)

CAUTION

On slippery surfaces, be prepared to counter nose-right rotation with left pedal as governor increases RPM.

NOTE

For hydraulic system check, use small cyclic inputs. With hydraulics OFF, there should be approximately one half inch of freeplay before encountering control stiffness and feedback. With hydraulics ON, controls should be free with no feedback or uncommanded motion.

NOTE

Idle mixture and speed may require adjustment as conditions vary from sea level standard. Refer to R44 Maintenance Manual for idle adjustment procedure.

NOTE

Before takeoff, pilot should uncover one ear and listen for any unusual noise which may indicate impending failure of a bearing or other component.

TAKEOFF PROCEDURE

1. Verify doors latched, governor and hydraulics ON, and RPM stabilized at 101 to 102%.
2. Clear area. Slowly raise collective until aircraft is light on skids. Reposition cyclic as required for equilibrium, then gently lift aircraft into hover.
3. Check gages in green, lower nose, and accelerate to climb speed following profile shown by height-velocity diagram in Section 5. Avoid exceeding two inches MAP above IGE hover power to prevent excessive nose-down attitude. If RPM drops below 101%, lower collective.

CRUISE

1. Verify RPM in green arc.
2. Set manifold pressure as desired with collective. Observe MAP and airspeed limits. Maximum recommended cruise speed is 110 KIAS.
3. Verify gages in green, warning lights out.

CAUTION

Do not exceed 110 KIAS except in smooth air, and then only with caution. In turbulence, use lower airspeed. If turbulence is significant or becomes uncomfortable for the pilot, use 60 to 70 KIAS.

CAUTION

In-flight leaning with engine mixture control is not allowed. Mixture must be full rich during flight.

NOTE

When loaded near aft CG limit, slight yaw oscillation during cruise can be stopped by applying a small amount of left pedal.

DOORS-OFF OPERATION

Maximum airspeed with any door(s) off is 100 KIAS. Warn passengers to secure loose objects and to keep head and arms inside cabin to avoid high velocity airstream.

CAUTION

Ensure all seat belts are buckled during door-off flight. Rear seat bottoms may lift if not restrained and items in baggage compartments could be blown out.

CAUTION

Flight with left door(s) removed is not recommended. Loose objects exiting left doors may damage tail rotor.

PRACTICE AUTOROTATION - POWER RECOVERY

1. Lower collective to down stop and adjust throttle as required for small tachometer needle separation.

CAUTION

To avoid inadvertent engine stoppage, do not roll throttle to full idle. Roll throttle off smoothly only enough for a small visible needle split.

NOTE

Governor is inactive below 80% engine RPM regardless of governor switch position.

NOTE

When entering autorotation from above 6000 feet, reduce throttle slightly before lowering collective to prevent engine overspeed.

2. Adjust collective to keep rotor RPM in green arc and adjust throttle for small needle separation.
3. Keep airspeed 60 to 70 KIAS.
4. At about 40 feet AGL, begin cyclic flare to reduce rate of descent and forward speed.
5. At about 8 feet AGL, apply forward cyclic to level aircraft and raise collective to control descent. Add throttle if required to keep RPM in green arc.

CAUTION

Simulated engine failures require prompt lowering of collective to avoid dangerously low rotor RPM. Catastrophic rotor stall could occur if the rotor RPM ever drops below 80% plus 1% per 1000 feet of altitude.

PRACTICE AUTOROTATION - WITH GROUND CONTACT

If practice autorotations with ground contact are required for demonstration purposes, perform in same manner as power recovery autorotations except:

Prior to cyclic flare, roll throttle off into overtravel spring and hold against hard stop until autorotation is complete. (This prevents throttle correlator from adding power when collective is raised.)

Always contact ground with skids level and nose straight ahead.

NOTE

Have landing gear skid shoes inspected frequently when practicing autorotations with ground contact. Rapid wear of skid shoes may occur.

HYDRAULICS-OFF TRAINING

Hydraulic system failure may be simulated using the cyclic-mounted hydraulic switch.

CAUTION

With hydraulics switched OFF, controlling helicopter in a hover may be difficult due to control system feedback forces.

CAUTION

Before switching hydraulics from OFF to ON, relax force on cyclic and collective to avoid overcontrolling.

DESCENT, APPROACH, AND LANDING

1. Reduce power with collective as desired. Observe airspeed limits. Maximum recommended airspeed is 110 KIAS except in smooth air.

CAUTION

Do not initiate a descent with forward cyclic. This can produce a low-G condition. Always initiate a descent by lowering collective.

2. Make final approach into wind at lowest practical rate of descent with initial airspeed of 60 knots.
3. Reduce airspeed and altitude smoothly to hover. (Be sure rate of descent is less than 300 FPM before airspeed is reduced below 30 KIAS.)
4. From hover, lower collective gradually until ground contact.
5. After initial ground contact, lower collective to full down position.

CAUTION

When landing on a slope, return cyclic control to neutral before reducing rotor RPM.

CAUTION

Never leave helicopter flight controls unattended while engine is running.

CAUTION

Hold throttle closed if passenger is entering or exiting with engine running and left seat collective installed.

SHUTDOWN PROCEDURE

Collective down, RPM 60-70% Friction ON
Cyclic and pedals neutral Friction ON
CHT drop Throttle closed
Clutch switch Disengage
Wait 30 seconds Mixture OFF
Wait 30 seconds Apply rotor brake
Clutch light Extinguishes
Avionics, alt, battery, and ignition switches OFF

CAUTION

Do not slow rotor by raising collective during shutdown. Blades may flap and strike tailcone.

NOTE

During idle and after engine shutdown, pilot should uncover one ear and listen for unusual noise which may indicate impending failure of a bearing or other component.

NOTE

HYD switch should be left ON for startup and shutdown to reduce possibility of unintentional hydraulics-off liftoff. Switch OFF only for pre-takeoff controls check or hydraulics-off training.

NOTE

Rotor brake should be left engaged after shutdown to disable starter buttons and reduce possibility of unintentional starter engagement.

NOISE ABATEMENT

To improve the quality of our environment and to dissuade overly restrictive ordinances against helicopters, it is imperative that every pilot minimize noise irritation to the public. Following are several techniques which should be employed when possible.

1. Avoid flying over outdoor assemblies of people. When this cannot be avoided, fly as high as practicable, preferably over 2000 feet AGL.
2. Avoid blade slap. Blade slap generally occurs at airspeeds below 100 KIAS. It can usually be avoided by maintaining 100 KIAS until rate of descent is over 1000 FPM, then using a fairly steep approach until airspeed is below 65 KIAS. With the right door vent open, the pilot can easily determine those flight conditions which produce blade slap and develop piloting techniques to eliminate or reduce it.
3. When departing from or approaching a landing site, avoid prolonged flight over noise sensitive areas. Always fly above 500 feet AGL and preferably above 1000 feet AGL.
4. Repetitive noise is far more irritating than a single occurrence. If you must fly over the same area more than once, vary your flight path to not overfly the same buildings each time.
5. When overflying populated areas, look ahead and select the least noise sensitive route.

NOTE

Above procedures do not apply where they would conflict with Air Traffic Control clearances or instructions or when, in the pilot's judgment, they would result in an unsafe flight path.